

Polyimide resins and their carbon fiber reinforced composites

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Carbon fiber reinforced polyimide composites, due to their unique combination of thermal and mechanical properties, have been widely used in recent years.^[1] The method to obtain polyimides composites with good processability and toughness has always been a scientific and technical challenge. PETI-5 composite is the most outstanding one which has a CAI value > 300MPa.^[2] There are still some improvements needed for this material. In order to fabricate composites with better quality, a resin with better melt flow is required. Moreover, a precursor solution with high solid content, low viscosity, and solvents which can be easily removed is also needed.^[3] These requirements call for more challenges for the fabrication of high performance polyimide composites.

In present work, a series of biphenyl-type imide oligomers end-capped by 4-phenylethynyl phthalic anhydride (4-PEPA) were synthesized through a modified PMR route. Stable resin solutions with solid content of 50% and viscosity of 40-50 cP were obtained. The solvents and volatiles of the resins were easily removed at low temperature as shown in Fig.1. B-stage resins with different copolymerization ratios and molecular weights were discussed by TGA, DSC and FTIR.

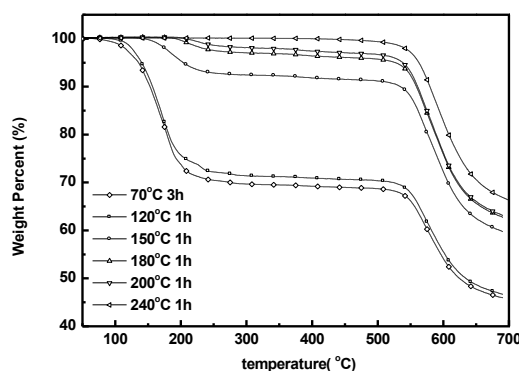


Fig.1 TGA curves of PI-1 after different thermal treatments

Different mole ratios of a-BPDA were incorporated into the oligomer backbones, and the processibility was significantly improved when the a-BPDA content reached 20% (PI-3) or higher (Fig. 2). This result should be attributed to the asymmetric structure of a-BPDA. After cured, PI-3 and PI-4, which had a molecular weight of 5000 g/mol, exhibited excellent mechanical properties as shown in Table 1. PI-3, which had the best overall properties with tensile strength >120MPa, flexural strength >150MPa and elongation at break >18%, was fabricated into carbon fiber laminates.

Table 1 Characterizations and properties of cured polyimide resins

Samples	TGA			DMA		Tensile properties			Flexural properties	
	T _d (°C)	T ₅ (°C)	R ₇₀₀ (%)	E' (°C)	tan δ (°C)	Strength (MPa)	Modulus (GPa)	Elongation (%)	Strength (MPa)	Modulus (GPa)
PI-3	546	548	69.3	273	283	124.3	2.0	18.8	154.8	3.1
PI-4	538	540	69.8	270	286	124.8	2.0	16.2	157.0	3.2

Through an optimized curing procedure, unidirectional and quasi-isotropic laminates were

fabricated with good quality. After impacted at an energy of 6.7 KJ/m, only a small damage could be observed in the center of the laminate indicated by C-scan (Fig.3). High retention of mechanical properties at elevated temperature (~250°C) and high damage tolerance were obtained on the composites laminates as shown in Table 2. These properties indicated a promising potential for future aerospace application.

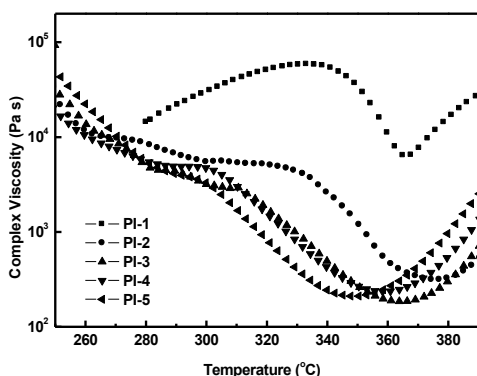


Fig.2 Rheology behaviors

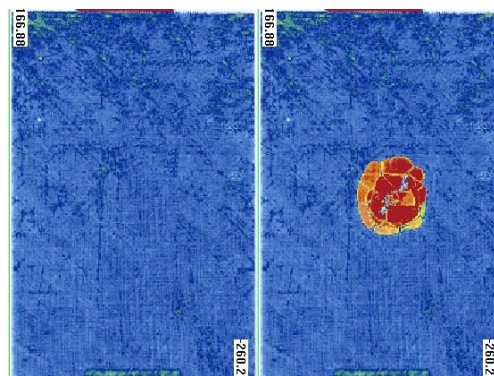


Fig.3 C-scan images before and after impact

Table 2 Mechanical properties of the laminates

	Lay-up	Properties
0° Flexural strength, MPa		
RT	[0°] ₁₂	1536.8 ± 86.7
177 °C		1137.9 ± 71.8
250 °C		819.1 ± 36.2
0° Flexural modulus, GPa		
RT	[0°] ₁₂	143.8 ± 4.6
177 °C		144.4 ± 3.3
250 °C		140.0 ± 0.9
CAI strength, MPa	[-45°,0°,45°,90°] _{3s}	229.2 ± 13.2
OHC strength, MPa	[-45°,0°,45°,90°] _{2s}	303.2 ± 17.5
OHT strength, MPa	[-45°,0°,45°,90°] _{2s}	366.9 ± 49.9

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